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| APPLICATION NO. | FILING DATE | FIRST NAMED INVENTOR | ATTORNEY DOCKET NO. | CONFIRMATION NO. |
|-----------------|-------------|----------------------|---------------------|------------------|
| 10/074,736      | 10/29/2001  | John M. Robertson    | F-7561              | 1087             |

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[REDACTED]

PITTMAN, ZIDIA T

| ART UNIT | PAPER NUMBER |
|----------|--------------|
| 1725     | 3            |

DATE MAILED: 01/02/2003

Please find below and/or attached an Office communication concerning this application or proceeding.

|                              |                           |                  |   |
|------------------------------|---------------------------|------------------|---|
| <b>Office Action Summary</b> | Application No.           | Applicant(s)     | D |
|                              | 10/074,736                | ROBERTSON ET AL. |   |
|                              | Examiner<br>Zidia Pittman | Art Unit<br>1725 |   |

-- The MAILING DATE of this communication appears on the cover sheet with the correspondence address --

#### Period for Reply

A SHORTENED STATUTORY PERIOD FOR REPLY IS SET TO EXPIRE 3 MONTH(S) FROM THE MAILING DATE OF THIS COMMUNICATION.

- Extensions of time may be available under the provisions of 37 CFR 1.136(a). In no event, however, may a reply be timely filed after SIX (6) MONTHS from the mailing date of this communication.
- If the period for reply specified above is less than thirty (30) days, a reply within the statutory minimum of thirty (30) days will be considered timely.
- If NO period for reply is specified above, the maximum statutory period will apply and will expire SIX (6) MONTHS from the mailing date of this communication.
- Failure to reply within the set or extended period for reply will, by statute, cause the application to become ABANDONED (35 U.S.C. § 133).
- Any reply received by the Office later than three months after the mailing date of this communication, even if timely filed, may reduce any earned patent term adjustment. See 37 CFR 1.704(b).

#### Status

- 1) Responsive to communication(s) filed on 29 October 2001.
- 2a) This action is FINAL.                  2b) This action is non-final.
- 3) Since this application is in condition for allowance except for formal matters, prosecution as to the merits is closed in accordance with the practice under *Ex parte Quayle*, 1935 C.D. 11, 453 O.G. 213.

#### Disposition of Claims

- 4) Claim(s) 1-20 is/are pending in the application.
- 4a) Of the above claim(s) 7 and 15-20 is/are withdrawn from consideration.
- 5) Claim(s) \_\_\_\_\_ is/are allowed.
- 6) Claim(s) 1-6 and 8-14 is/are rejected.
- 7) Claim(s) \_\_\_\_\_ is/are objected to.
- 8) Claim(s) \_\_\_\_\_ are subject to restriction and/or election requirement.

#### Application Papers

- 9) The specification is objected to by the Examiner.
- 10) The drawing(s) filed on \_\_\_\_\_ is/are: a) accepted or b) objected to by the Examiner.  
Applicant may not request that any objection to the drawing(s) be held in abeyance. See 37 CFR 1.85(a).
- 11) The proposed drawing correction filed on \_\_\_\_\_ is: a) approved b) disapproved by the Examiner.  
If approved, corrected drawings are required in reply to this Office action.
- 12) The oath or declaration is objected to by the Examiner.

#### Priority under 35 U.S.C. §§ 119 and 120

- 13) Acknowledgment is made of a claim for foreign priority under 35 U.S.C. § 119(a)-(d) or (f).
- a) All b) Some \* c) None of:
1. Certified copies of the priority documents have been received.
  2. Certified copies of the priority documents have been received in Application No. \_\_\_\_\_.
  3. Copies of the certified copies of the priority documents have been received in this National Stage application from the International Bureau (PCT Rule 17.2(a)).
- \* See the attached detailed Office action for a list of the certified copies not received.
- 14) Acknowledgment is made of a claim for domestic priority under 35 U.S.C. § 119(e) (to a provisional application).
- a) The translation of the foreign language provisional application has been received.
- 15) Acknowledgment is made of a claim for domestic priority under 35 U.S.C. §§ 120 and/or 121.

#### Attachment(s)

- |  |  |
|--|--|
| <input checked="" type="checkbox"/> Notice of References Cited (PTO-892)                             | <input type="checkbox"/> Interview Summary (PTO-413) Paper No(s). _____  |
| <input type="checkbox"/> Notice of Draftsperson's Patent Drawing Review (PTO-948)                    | <input type="checkbox"/> Notice of Informal Patent Application (PTO-152) |
| <input checked="" type="checkbox"/> Information Disclosure Statement(s) (PTO-1449) Paper No(s) _____ | <input type="checkbox"/> Other: _____                                    |

**DETAILED ACTION**

***Election/Restrictions***

Restriction to one of the following inventions is required under 35 U.S.C. 121:

- I. Claims 1-6 and 8-20, drawn to method of repair, classified in class 228, subclass 119.
- II. Claim 7, drawn to a part, classified in class 428, subclass 544+.

The inventions are distinct, each from the other because of the following reasons:

Inventions I and II are related as process of making and product made. The inventions are distinct if either or both of the following can be shown: (1) that the process as claimed can be used to make other and materially different product or (2) that the product as claimed can be made by another and materially different process (MPEP § 806.05(f)). In the instant case the product as claimed can be produced by a different process, such as diffusion bonding or welding.

Because these inventions are distinct for the reasons given above and have acquired a separate status in the art as shown by their different classification, restriction for examination purposes as indicated is proper.

***SPECIES ELECTION***

This application contains claims directed to the following patentably distinct species of the claimed invention:

If the applicant elects Group I, an election of Species (I-II) is required.

SPECIES I (claims 1-6 and 8-14) drawn to a method of repairing a part including providing a wrought part having a contact area and an anomaly that renders the part unsuitable.

SPECIES II (claims 15-20) drawn to a method of repairing a rotating disk or drum rotor or a gas turbine engine including providing a rotating disk or drum rotor made from a wrought material and having an arrangement of lugs and slots, at least one of said lugs or said slots having an anomaly thereon, pressing the contact area of the material against the contact area of the component so that the material bonds to the component, and treating the material to provide a desired shape to the disk or drum.

Applicant is required under 35 U.S.C. 121 to elect a single disclosed species for prosecution on the merits to which the claims shall be restricted if no generic claim is finally held to be allowable. Currently, no claim is generic.

Applicant is advised that a reply to this requirement must include an identification of the species that is elected consonant with this requirement, and a listing of all claims readable thereon, including any claims subsequently added. An argument that a claim is allowable or that all claims are generic is considered nonresponsive unless accompanied by an election.

Upon the allowance of a generic claim, applicant will be entitled to consideration of claims to additional species which are written in dependent form or otherwise include all the limitations of an allowed generic claim as provided by 37 CFR 1.141. If claims are added after the election, applicant must indicate which are readable upon the elected species. MPEP § 809.02(a).

Should applicant traverse on the ground that the species are not patentably distinct, applicant should submit evidence or identify such evidence now of record showing the species to be obvious variants or clearly admit on the record that this is the case. In either instance, if the examiner finds one of the inventions unpatentable over the prior art, the evidence or admission may be used in a rejection under 35 U.S.C. 103(a) of the other invention.

During a telephone conversation with Brian Hamilla on December 16, 2002 a provisional election was made with traverse to prosecute the invention of Group I, claims 1-6 and 8-20. Furthermore, based on the species restriction, a provisional election was made with traverse to prosecute the invention of Species I, claims 1-6 and 8-14. Affirmation of this election must be made by applicant in replying to this Office action. Claims 7 and 15-20 are withdrawn from further consideration by the examiner, 37 CFR 1.142(b), as being drawn to a non-elected invention.

Applicant is reminded that upon the cancellation of claims to a non-elected invention, the inventorship must be amended in compliance with 37 CFR 1.48(b) if one or more of the currently named inventors is no longer an inventor of at least one claim remaining in the application. Any amendment of inventorship must be accompanied by a request under 37 CFR 1.48(b) and by the fee required under 37 CFR 1.17(i).

***Claim Rejections - 35 USC § 102***

The following is a quotation of the appropriate paragraphs of 35 U.S.C. 102 that form the basis for the rejections under this section made in this Office action:

A person shall be entitled to a patent unless –

(b) the invention was patented or described in a printed publication in this or a foreign country or in public use or on sale in this country, more than one year prior to the date of application for patent in the United States.

Claims 1-14 are rejected under 35 U.S.C. 102(b) as being anticipated by Robertson et al (USPN 5,272,809).

Robertson et al teaches a technique for direct bonding cast and wrought materials. The present invention provides a method for joining a wrought superalloy to a cast superalloy material. The invention is exemplified by the repair of broken, damaged, or improperly machined parts of components for gas turbine engines, where such parts are made of high strength, high temperature materials such as nickel-base or cobalt-base superalloys, or titanium alloys, and also provides a method for the replacement of such parts when appropriate. Surprisingly, we have discovered that a wrought replacement part, such as a lug, may be joined to the cast vane assembly, by a forge joining technique. By use of a wrought replacement piece in a re-designed configuration, superior high temperature lug strength properties may be attained in the component, resulting in a vane assembly having adequate creep capability to meet design requirements, and superior high temperature strength in the lug per se. In accordance with an exemplary use of this invention, a cast vane assembly comprises a vane outside diameter platform, vanes, and an annulus inside diameter platform. The outer diameter platform has projecting lugs, adapted to fit or engage circumferential flanges or retaining rings on the engine case. Frequently, as a result of damage during installation or in use, lugs on such vane assembly are broken off, which illustrates a vane assembly having stubs, where retaining lugs were formerly located. These stubs, or damaged areas of the vane assembly, may be machined away, to leave a vane

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assembly having a uniform repair site, in the location of the failed or broken lug. An oversized retaining lug blank, having the requisite dimensions to replace the failed lug, is then positioned over the uniform repair site, in contact therewith. This replacement lug blank is preferably of the same general composition as the cast vane assembly composition, but of a wrought alloy corresponding thereto. The replacement blank is accurately clamped in position in relation to the repair site, and is forge joined thereto by heating the interface surfaces of the blank and the repair site to a temperature within the solution temperature range of both alloys, but not exceeding the solution heat treatment temperature of either alloy, and without causing local melting or significant grain growth in the vane assembly and the replacement lug blank. As previously indicated, the present invention is specifically exemplified by the repair of damaged gas turbine engine vane assemblies, which assemblies are normally made of a cast high temperature material, such as either non-hardenable or precipitation-hardenable nickel-base superalloys, or titanium. Preparation for a part repair includes machining away the broken area of the failed part. The replacement blank is sized to provide the desired part size and configuration after machining, and is configured so as to provide a bonding surface having a close tolerance fit to the bonding surface of the assembly. After the replacement blank is positioned in close proximity to the bonding surface of the assembly, as in a tooling fixture or jig, a forge joining pressure is applied to the blank. While this pressure is applied, the interface between the blank and the assembly bonding surface is locally heated. (column 4 line 15 – column 6 line 48)

With respect to the limitation requiring the part to not be fusion weldable, the examiner contends that although the cited reference doesn't explicitly state this limitation, it would be inherent that the materials utilized in the reference would have this same feature, since the materials of the claimed invention (nickel-based superalloy) corresponding to one of the cited materials of the reference.

Claims 1-14 are rejected under 35 U.S.C. 102(b) as being anticipated by Bogard et al (USPN 5,205,465).

Bogard et al teaches a method for replacing worn airseal lands on engine compressor or turbine disks. The present invention provides a method for the repair of broken, damaged, improperly machined, or worn airseal lands on a compressor or turbine disk, where such lands are made of high strength, high temperature materials such as nickel-base superalloys, or titanium alloys, and also provides a method for the replacement of such lands when appropriate. While the figures are illustrative of a turbine hub, it should be noted that the present invention is equally applicable to compressor disks, turbine disks, etc., wherein the central portion of the rotor is planar. The repair materials should be selected so as to form a strong solid state bond with the material of the disk or hub material. The preferred material would be the same material as the base material of the disk or hub. Figure 5 illustrates the disk assembly of Figure 2, with the worn airseal lands removed, leaving repair site stubs. These stubs should be preferably machined to a relatively uniform smoothness. Figure 6 illustrates a replacement ring to be bonded to the stubs. This replacement ring comprises an oversized load bearing upper section, a lower section, sized to match the stub of the

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original airseal land which is to be replaced, and a bonding surface. The purpose of the oversized area is to enable one to localize pressure in the narrower region so as to achieve a greater degree of upset at bonding surface. The replacement ring is placed in location on the appropriate stub of the inner airseal land or outer airseal land, and forge-joined thereto to yield a bonded disk and airseal land assembly which also illustrates the upset at the interfacial bond line. The newly bonded oversized replacement airseal land may now be heat treated, using a localized heat treatment to assure the airseal land interfacial bond joint having the desired microstructure and properties. After heat treatment, the new airseal land ring is now ready for machining to the final configuration wherein replacement airseal lands have been bonded to disk and machined to the appropriate configuration for use. As previously indicated, the replacement ring is bonded to the stubs by forge-joining. It is to be understood that Figure 7 is meant to represent a typical appropriate press. The term 'press' as used in this specification and appended claims is intended to encompass any device for forcing the components together under very high pressures and temperatures. The components are heated within the press to a temperature sufficient to put at least the replacement ring in a superplastic or highly deformable condition. Sufficient pressure is then applied to the components to cause the replacement ring to deform at its bonding surface to create line-on-line contact over the entire bonding surface of the replacement ring and the stub to which it is being applied. The high pressure and temperature are held until a solid state diffusion bond is achieved. (column 3 line 10 – column 4 line 47)

With respect to the limitation requiring the part to not be fusion weldable, the examiner contends that although the cited reference doesn't explicitly state this limitation, it would be inherent that the materials utilized in the reference would have this same feature, since the materials of the claimed invention (nickel-based superalloy or titanium alloy) corresponding to one of the cited materials of the reference.

### ***Conclusion***

The prior art made of record and not relied upon is considered pertinent to applicant's disclosure.

DeMichael et al (USPN 5,109,606) and Hyzak et al (USPN 5,106,012) are cited as of interest.

Any inquiry concerning this communication or earlier communications from the examiner should be directed to Zidia Pittman whose telephone number is (703) 305-1248. The examiner can normally be reached on Monday – Thursday and alternate Fridays from 8:30 am to 6:00 pm.

If attempts to reach the examiner by telephone are unsuccessful, the examiner's supervisor, Tom Dunn, can be reached at (703) 308-3318. The official fax phone number for the organization where this application or proceeding is assigned is (703) 305-7718. The unofficial fax number for art unit 1725 is (703) 305-6078.

Any inquiry of a general nature or relating to the status of this application or proceeding should be directed to the receptionist whose telephone number is (703) 308-0661.

*ZP*  
12/20/02

*Tom Dunn*  
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SUPERVISORY PATENT EXAMINER  
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